



Accelerating PDE-Constrained Optimization using Adaptive Reduced-Order Models

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The desire to solve optimization problems governed by partial differential equations exists in all fields of science and engineering. These PDE-constrained optimization problems inevitably require a large number of solutions of the partial differential equation of interest and become prohibitively expensive if a fine discretization is required. We introduce a fast algorithm for solving such optimization problems that leverages adaptive reduced-order models and is provably globally convergent.

Motivation

- The problem of **designing a hull** to maximize energy absorption of the armor in order **protect the occupants** in the event of an underbody blast is of utmost importance
- Testing a given design incurs substantial cost (\$5M): turn to **computational tools** to **analyze and design** such systems
- Computational analysis of a single design may take **many hours on a supercomputer** due to the complex geometry and physics involved
- Optimization of such a system may require **simulation of thousands of designs**, rendering the problem practically infeasible



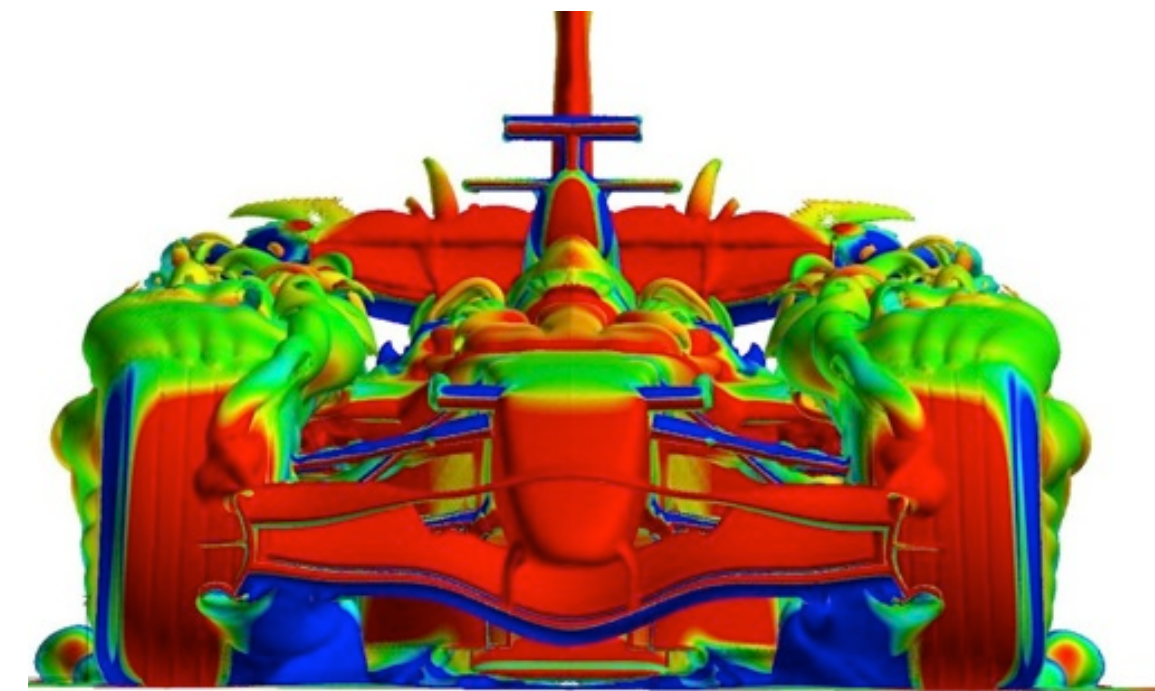
- We introduce a framework for solving such PDE-constrained optimization problems using **reduced-order models** with the goal of substantial CPU saving.

Goal

Accelerate solution of a PDE-constrained optimization problem using a Reduced-Order Model (ROM) as a surrogate for the PDE

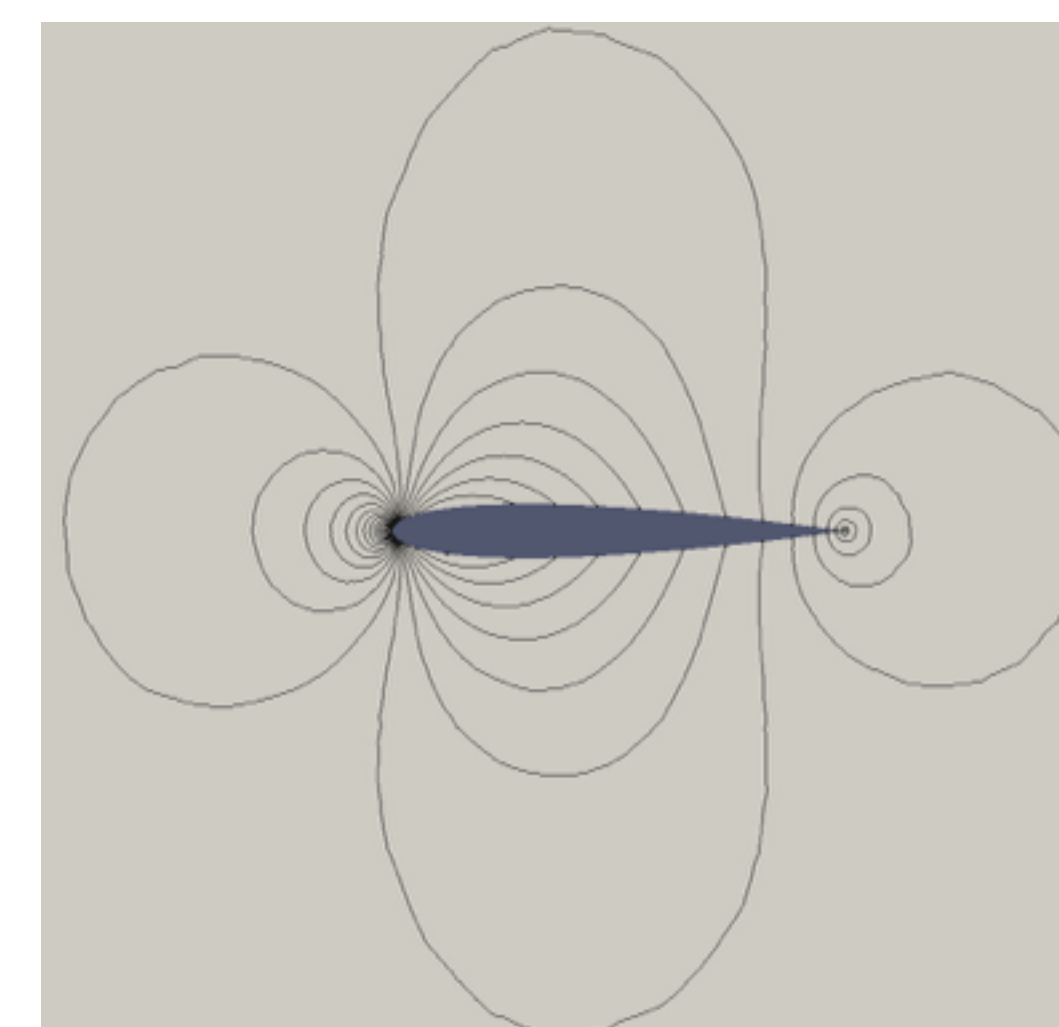
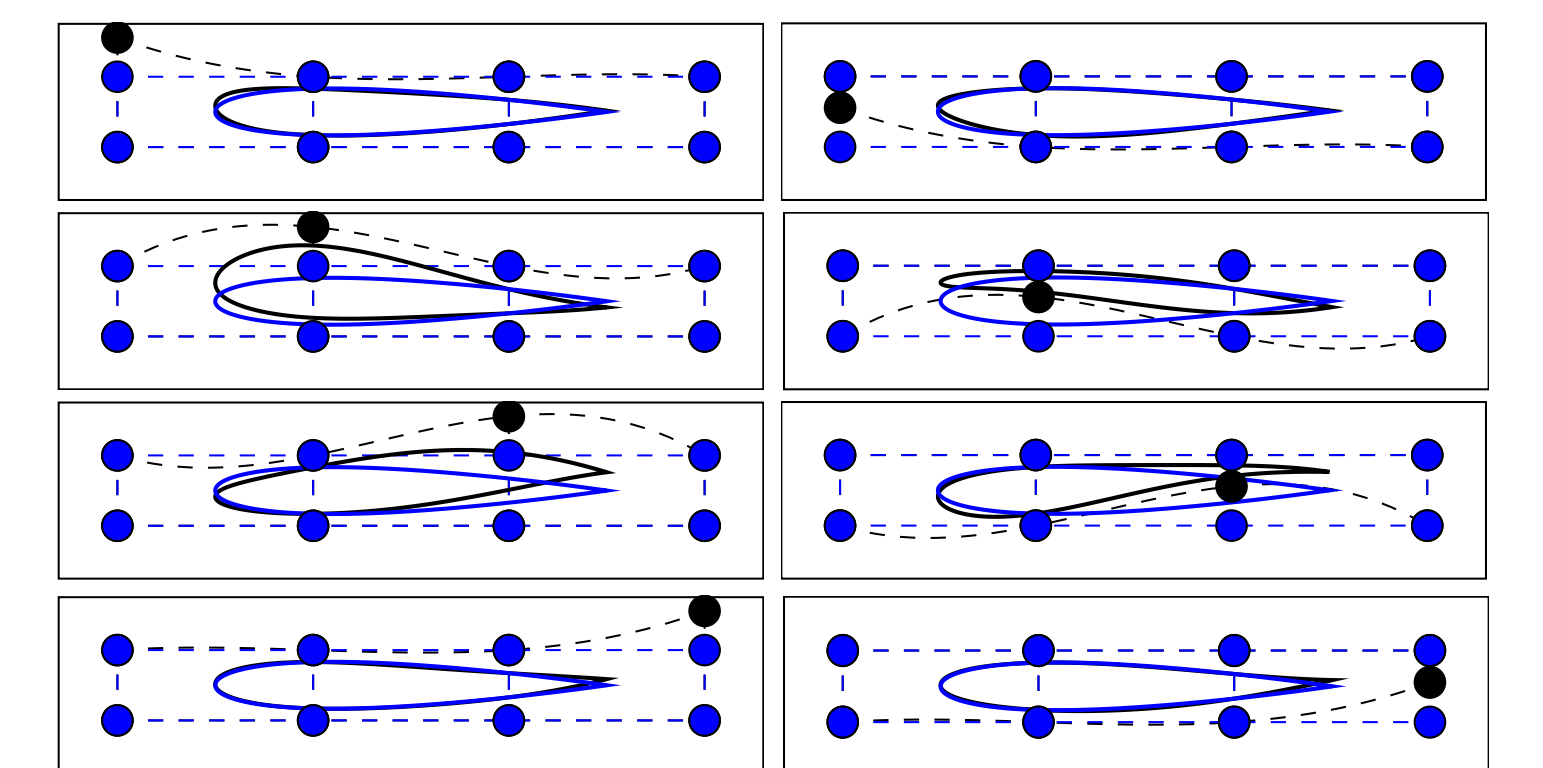
$$\begin{aligned} &\text{minimize}_{\mathbf{w} \in \mathbb{R}^N, \boldsymbol{\mu} \in \mathbb{R}^p} f(\mathbf{w}, \boldsymbol{\mu}) \\ &\text{subject to } \mathbf{R}(\mathbf{w}, \boldsymbol{\mu}) = 0 \end{aligned}$$

$\mathbf{R}(\mathbf{w}, \boldsymbol{\mu})$ — discretized PDE
 \mathbf{w} — state vector
 $\boldsymbol{\mu}$ — parameter

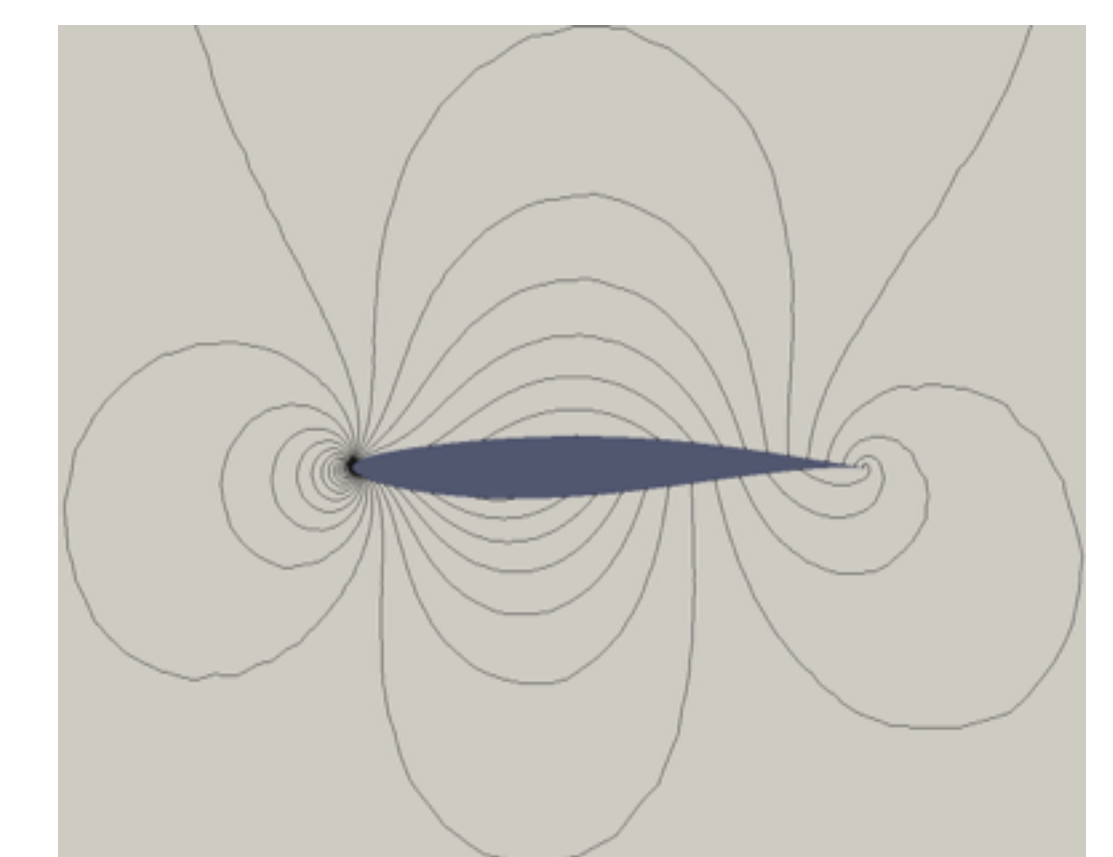


Aerodynamic Shape Optimization

In this section, the proposed optimization algorithm that leverages adaptive reduced-order models is compared to a standard technique for PDE-constrained optimization on the problem of recovering a NACA0012 geometry from a NACA0012 geometry by considering only the discrepancy in the pressure distribution.

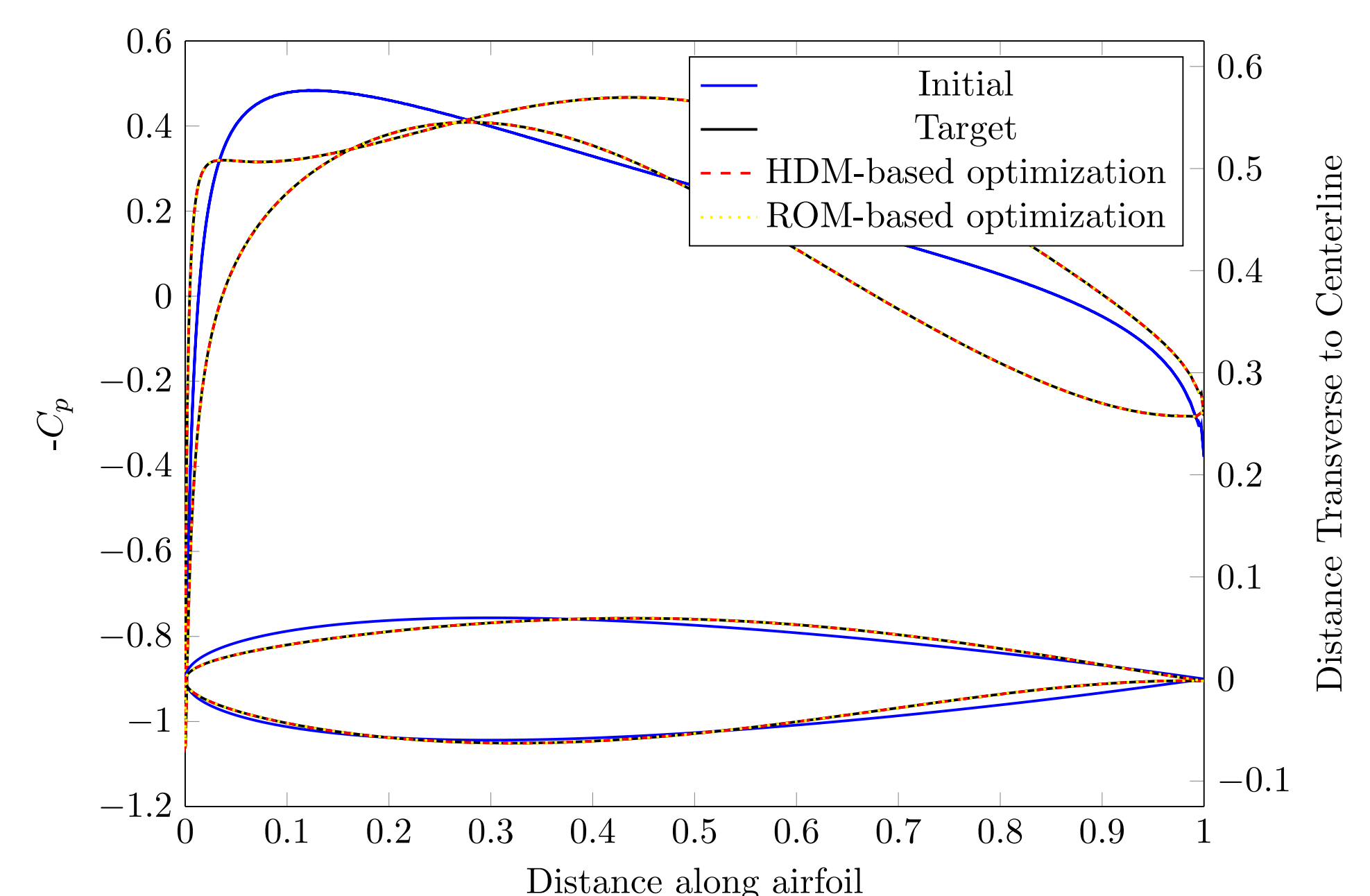


NACA0012
(M = 0.5, AOA = 0)



RAE2822
(M = 0.5, AOA = 0)

Initial/Target shape and pressure distribution with optimization results



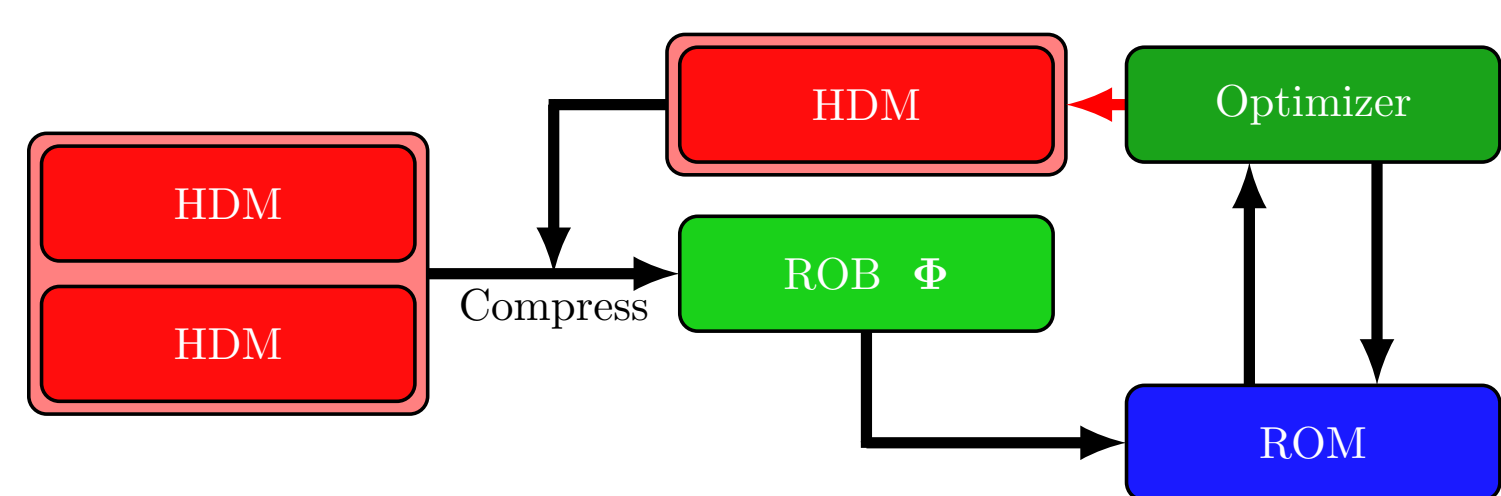
Optimization via Adaptive Reduced-Order Model

Assume state vector lies in r -dimensional *trial* subspace where $r \ll N$, defined by the Reduced Basis (RB) and project equations into r -dimensional *test* subspace.

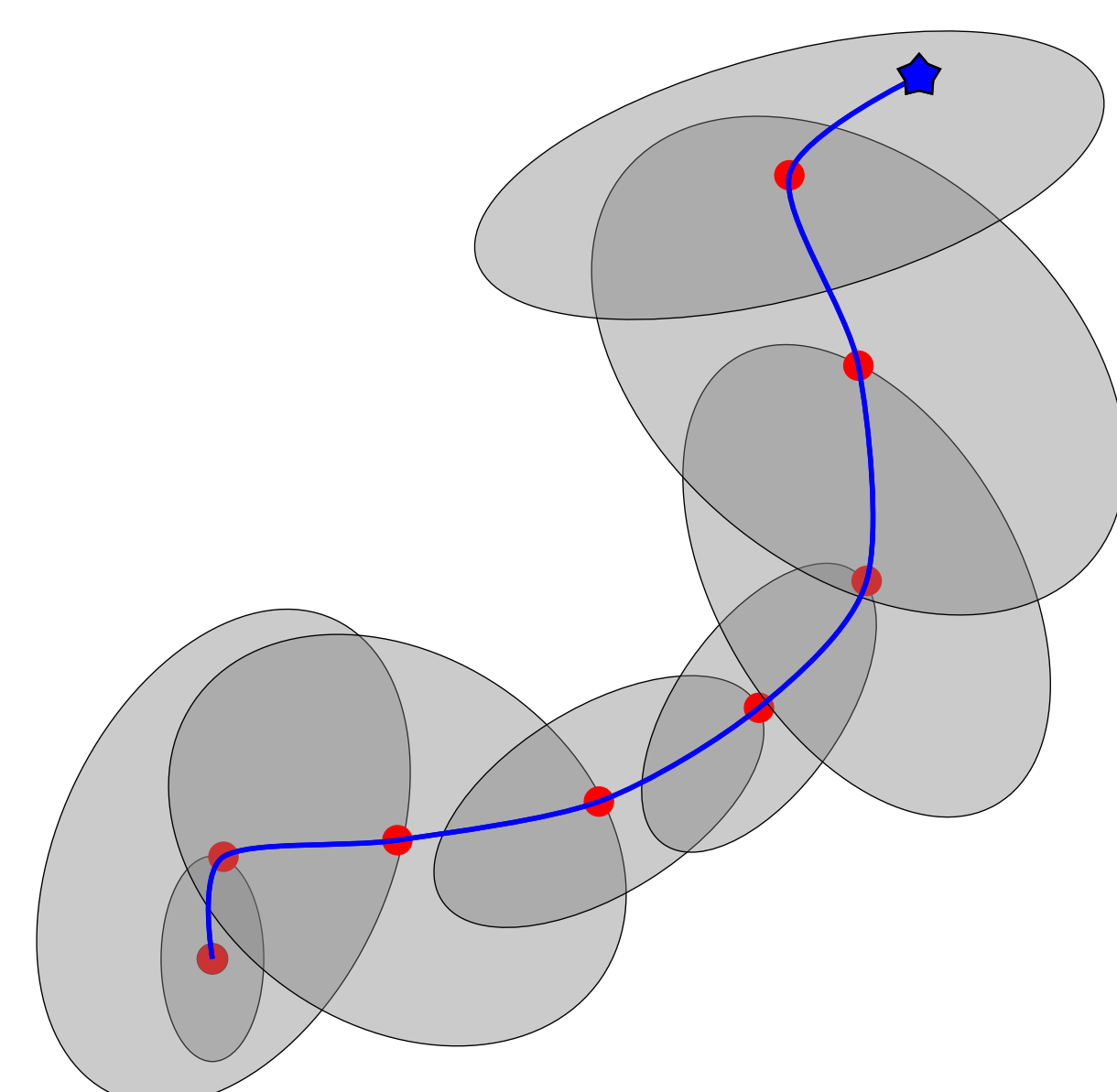
$$\mathbf{w} = \boldsymbol{\Phi} \mathbf{y} \quad \boldsymbol{\Psi}^T \mathbf{R}(\boldsymbol{\Phi} \mathbf{y}, \boldsymbol{\mu}) = 0$$

Non-quadratic trust-region model problem: *error-aware ROM-constrained optimization*

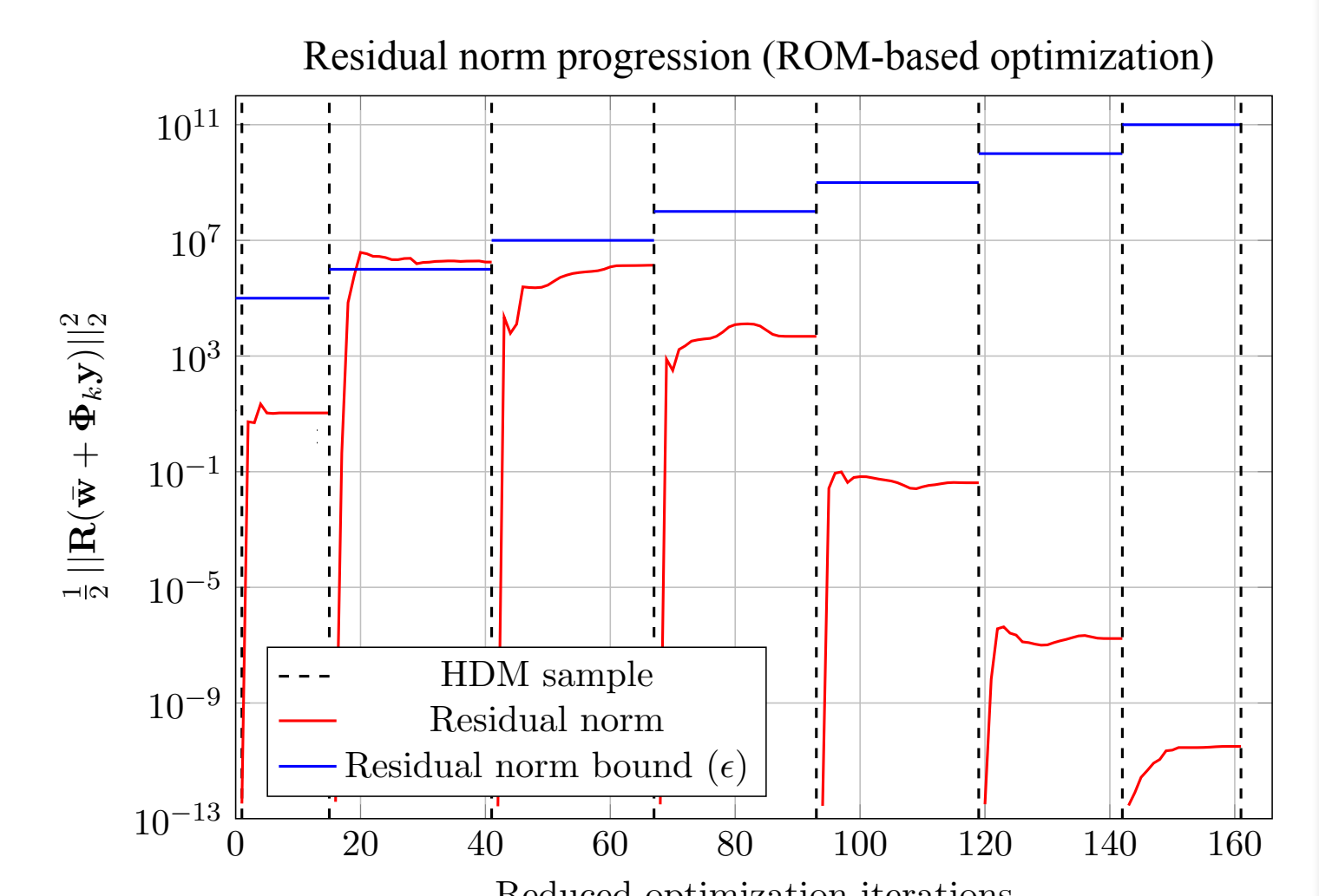
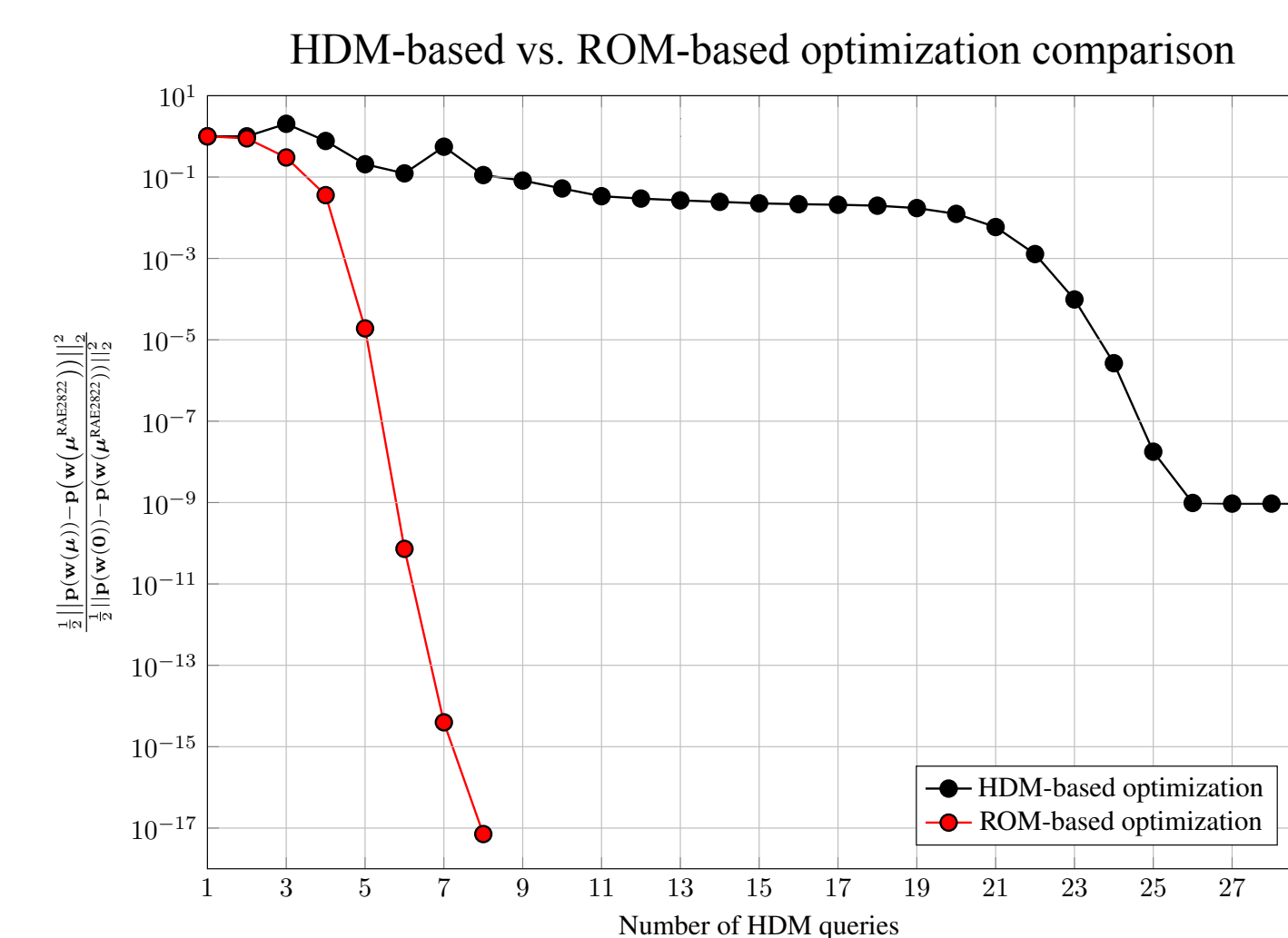
$$\begin{aligned} &\text{minimize}_{\mathbf{y} \in \mathbb{R}^r, \boldsymbol{\mu} \in \mathbb{R}^p} f(\boldsymbol{\Phi}_k \mathbf{y}, \boldsymbol{\mu}) \\ &\text{subject to } \boldsymbol{\Psi}_k^T \mathbf{R}(\boldsymbol{\Phi}_k \mathbf{y}, \boldsymbol{\mu}) = 0 \\ &\quad \|\mathbf{R}(\boldsymbol{\Phi}_k \mathbf{y}, \boldsymbol{\mu})\| \leq \Delta_k \end{aligned}$$



Workflow Schematic

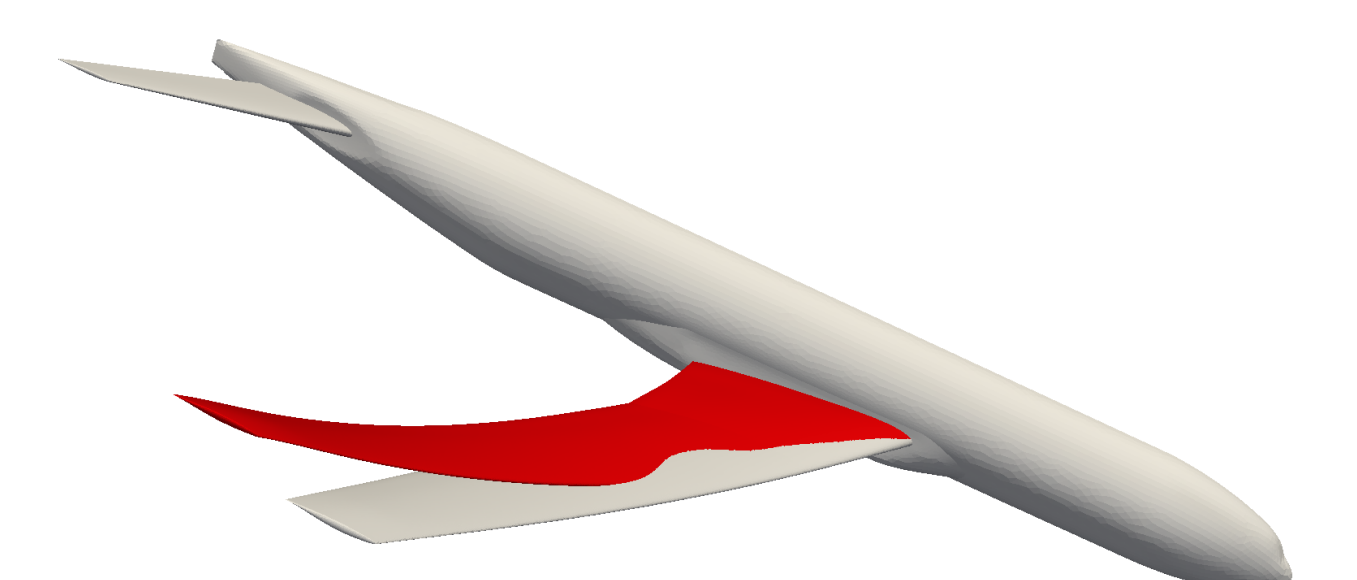


Parameter Space



Conclusion

- Introduced new globally convergent method for accelerating PDE-constrained optimization using Reduced-Order Models.
- Factor of 4 fewer HDM queries** observed on aerodynamic shape optimization problem where the optimal solution was recovered to machine precision.
- Ongoing work is focused on demonstrating the proposed approach on a large-scale problem - design of the Common Research Model.



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Global Convergence Theorem

Define the implicit functions

$$g(\boldsymbol{\mu}) := f(\mathbf{w}(\boldsymbol{\mu}), \boldsymbol{\mu}) \quad m_k(\boldsymbol{\mu}) := f(\boldsymbol{\Phi}_k \mathbf{y}(\boldsymbol{\mu}), \boldsymbol{\mu})$$

and assume they are continuously differentiable with bounded Hessian. If the following relaxed first-order conditions are met (guaranteed by the proposed minimum-residual primal and sensitivity reduced-order model framework): $\exists \xi > 0$

$$m_k(\boldsymbol{\mu}_k) = g(\boldsymbol{\mu}_k) \quad \|\nabla g(\boldsymbol{\mu}_k) - \nabla m_k(\boldsymbol{\mu}_k)\| \leq \xi \min\{\|\nabla m_k(\boldsymbol{\mu}_k)\|, \Delta_k\}$$

then the proposed trust-region algorithm produces a sequence of iterates that satisfies

$$\liminf_{k \rightarrow \infty} \|\nabla m_k(\boldsymbol{\mu}_k)\| = \liminf_{k \rightarrow \infty} \|\nabla g(\boldsymbol{\mu}_k)\| = 0$$

Thus the algorithm converges to a local minimum from *any starting point*.

Collaborations

The following collaboration efforts are planned:

ARL/CSD: Pat Collins, on the CFD ROM component and its introduction at ARL/VTD where AERO-F is now known. Anticipated applications are design optimization of MAVs and flapping wings, among others

TARDEC: Matt Castanier, on the structural dynamics ROM component with applications to armor design optimization